

Professional Propulsion Systems

SYSTEM SPECIFICATIONS

ENGINE



Name:	4202HF
Manufacturer:	HIRTH ENGINES
Type:	2-cylinder
Displacement:	183 cm³
Max. performance:	11 kW at 6500 RPM
Weight:	11.5 kg
RPM range:	2500–6500 RPM
Running direction:	Clockwise

PROPELLER



Name:	32x12 2B CCW (Direction guide)
Manufacturer:	Mejzlik
Diameter:	32 in
Pitch:	12 in
Mass:	340 g
Contact:	info@mejzlik.eu

ANALYSIS



Need expert guidance on analyzing your flight performance?

Our team provides a comprehensive analysis of RPM calculations, motor and propeller efficiency, including customized propeller selection recommendations to ensure your system achieves maximum efficiency.

Please reach out to us at info@mejzlik.eu or info@hirthengines.com

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PERFORMANCE OF THE SYSTEM

Flight velocity

0 m/s

Rotational Speed [RPM]	Thrust [N]	Torque [Nm]	Mechanical Power [W]	Propulsion efficiency [%]
2500	95	4.14	1083	0
2750	116	5.00	1439	0
3000	139	6.00	1886	0
3250	163	7.09	2414	0
3500	191	8.32	3050	0
3750	221	9.65	3788	0
4000	253	11.12	4659	0
4250	291	12.71	5657	0
4500	324	14.39	6782	0
4750	369	16.22	8068	0
5000	414	18.05	9449	0
5250	452	20.62	11338	0
5500	510	22.92	13201	0
5750	557	25.44	15321	0
6000	616	28.51	17911	0
6250	677	31.86	20855	0
6500	741	35.62	24242	0

Flight velocity

10 m/s

Rotational Speed [RPM]	Thrust [N]	Torque [Nm]	Mechanical Power [W]	Propulsion efficiency [%]
2500	65	4.19	1096	60%
2750	84	5.17	1490	56%
3000	105	6.26	1965	53%
3250	128	7.44	2533	51%
3500	154	8.73	3200	48%
3750	181	10.12	3975	46%
4000	212	11.65	4879	43%
4250	244	13.29	5913	41%
4500	279	15.06	7099	39%
4750	317	16.99	8449	38%
5000	358	19.07	9987	36%
5250	401	21.34	11730	34%
5500	448	23.80	13707	33%
5750	500	26.49	15951	31%
6000	555	29.43	18492	30%
6250	614	32.65	21372	29%
6500	674	36.22	24653	27%

PERFORMANCE OF THE SYSTEM

Flight velocity

20 m/s

Rotational Speed [RPM]	Thrust [N]	Torque [Nm]	Mechanical Power [W]	Propulsion efficiency [%]
2500	18	2.48	650	54%
2750	33	3.41	983	67%
3000	51	4.55	1430	71%
3250	72	5.84	1987	72%
3500	94	7.23	2650	71%
3750	119	8.72	3425	70%
4000	146	10.33	4328	68%
4250	176	12.06	5366	66%
4500	209	13.92	6559	64%
4750	244	15.92	7917	62%
5000	283	18.07	9461	60%
5250	324	20.40	11213	58%
5500	369	22.91	13193	56%
5750	417	25.63	15432	54%
6000	469	28.59	17964	52%
6250	525	31.84	20840	50%
6500	585	35.40	24099	49%

Flight velocity

30 m/s

Rotational Speed [RPM]	Thrust [N]	Torque [Nm]	Mechanical Power [W]	Propulsion efficiency [%]
2500	18	2.48	650	54%
2750	33	3.41	983	67%
3000	51	4.55	1430	71%
3250	72	5.84	1987	72%
3500	94	7.23	2650	71%
3750	119	8.72	3425	70%
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5500	369	22.91	13193	56%
5750	417	25.63	15432	54%
6000	469	28.59	17964	52%
6250	525	31.84	20840	50%
6500	585	35.40	24099	49%

PERFORMANCE OF THE SYSTEM

Flight velocity

40 m/s

Rotational Speed [RPM]	Thrust [N]	Torque [Nm]	Mechanical Power [W]	Propulsion efficiency [%]
2500	-30	1.25	327	—
2750	-33	1.59	457	—
3000	-33	1.84	577	—
3250	-32	2.21	752	—
3500	-29	2.62	961	—
3750	-22	3.14	1235	—
4000	-11	4.02	1682	-26%
4250	4	5.06	2250	8%
4500	26	6.41	3022	35%
4750	52	7.87	3913	53%
5000	81	9.58	5017	65%
5250	114	11.82	6501	70%
5500	151	14.39	8287	73%
5750	190	17.22	10367	73%
6000	233	20.26	12731	73%
6250	281	23.59	15441	73%
6500	332	27.19	18507	72%

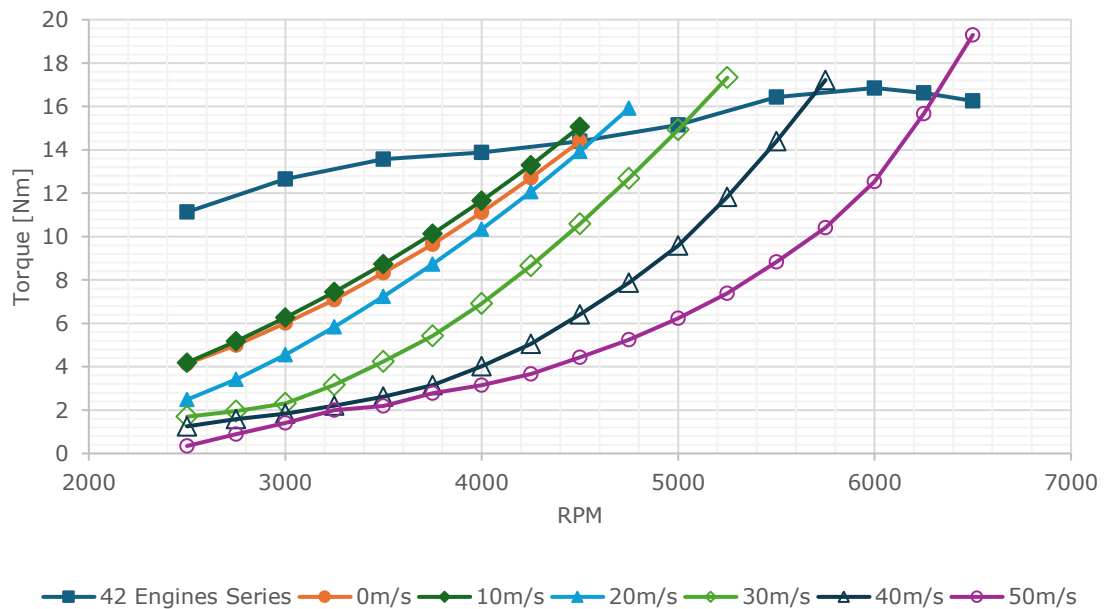
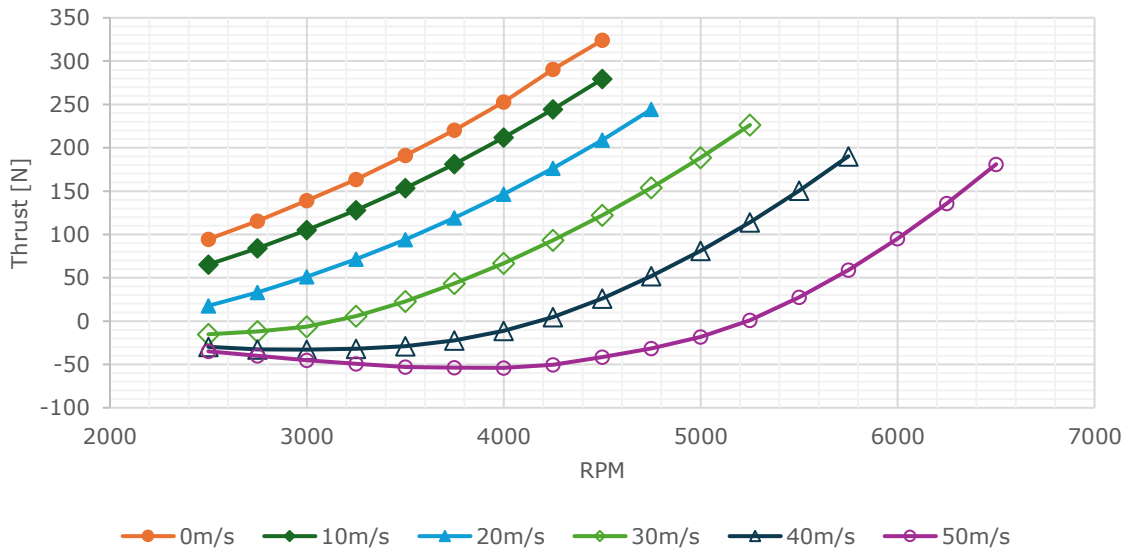
Flight velocity

50 m/s

Rotational Speed [RPM]	Thrust [N]	Torque [Nm]	Mechanical Power [W]	Propulsion efficiency [%]
2500	-35	0.34	90	—
2750	-40	0.89	255	—
3000	-45	1.41	443	—
3250	-49	2.00	679	—
3500	-53	2.19	803	—
3750	-54	2.77	1089	—
4000	-54	3.15	1321	—
4250	-50	3.67	1633	—
4500	-42	4.44	2091	—
4750	-31	5.24	2608	—
5000	-18	6.23	3262	-28%
5250	1	7.39	4063	1%
5500	27	8.83	5085	27%
5750	59	10.40	6265	47%
6000	95	12.53	7875	60%
6250	136	15.67	10255	66%
6500	181	19.30	13137	69%

PERFORMANCE OF THE SYSTEM

Hirth 42 series + Mejlzik 32x12 Performance in flight



NOTE



Data presented in this product sheet are a combination of measurements of engine performance (RPM, torque), which is complemented with propeller data, simulated in Mejlziks proprietary simulation software. The greyed out values are above engine limit.

Data is valid for 0m ISA. Propeller performance simulation accuracy can diverge at higher tip speeds (above 0.7 Mach). Propeller is structurally safe to operate below Mach 1 tip speed.

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