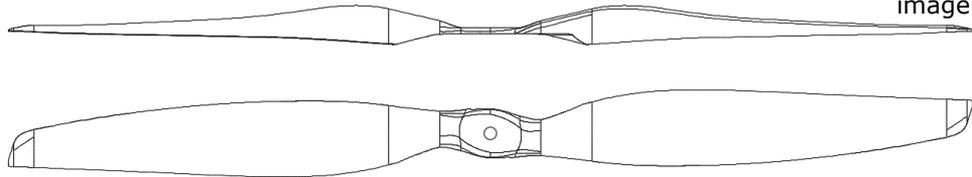


PROPELLER

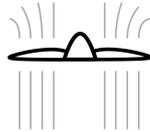
18x6 2B MC

PN: 2180600, 2180601

*Illustrative
image only



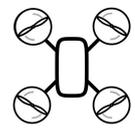
25 g
Mass



12.3 kgf
Max Thrust



18.0"
Diameter



Multicopter

Engine/Motor type: Electric

Rotation direction: Counter-clockwise and Clockwise available (Direction Guide)

Mass [g]: $25 \pm 10.0\%$

Moment of inertia [kgm^2]¹: $4.35\text{e-}04$

Core diameter [mm]: 16.9 (Drilling guide)

Limit hover RPM²: 10000

Limit forward speed [m/s]: Not calculated

Working temperature [°C]: from -20°C to 60°C

Production method: Wet layup

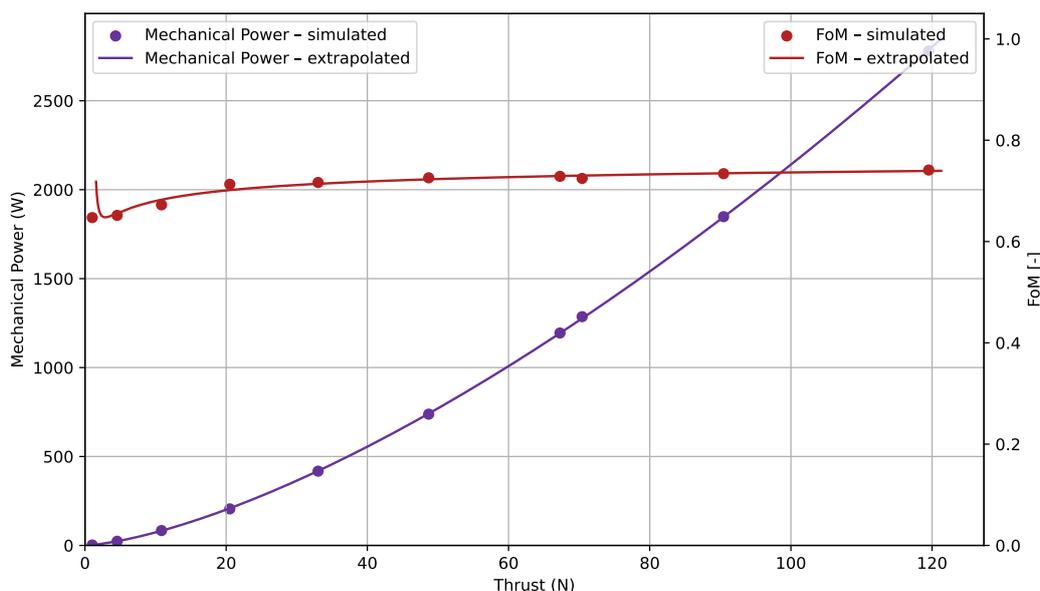
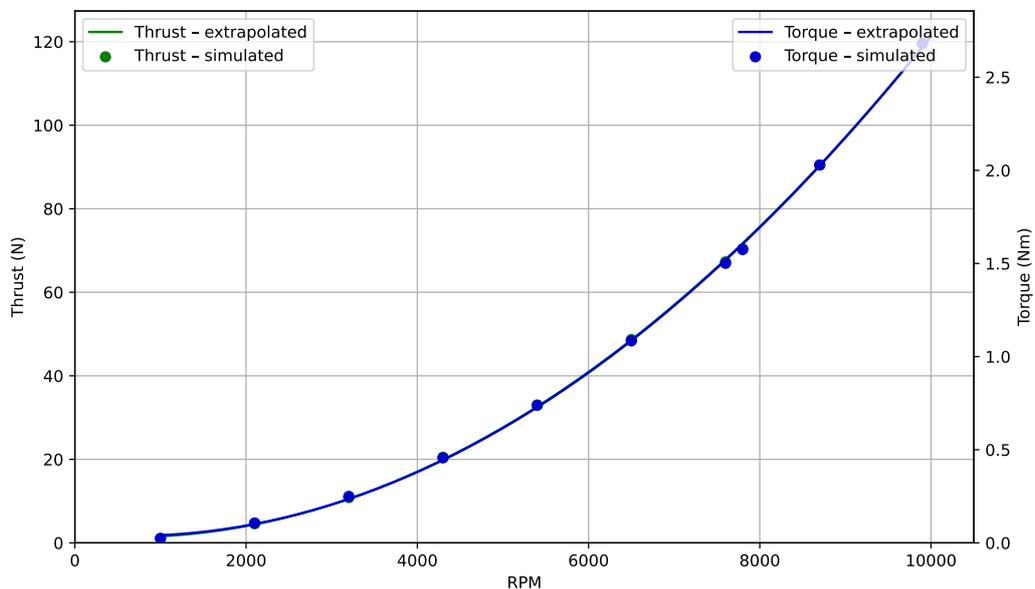
¹ Moment of inertia is only an estimation: $I = \frac{1}{24} \cdot \text{mass} \cdot \text{diameter}^2 \cdot n^{\circ} \text{of blades}$

² RPM is limited by tip speed; forward speed reduces the limit.

For more information or custom propeller options, contact info@mejlizk.eu.
Operation manual: Propeller Maintenance and Repair Manual

Static Performance Data

Simulated data



$$\text{Thrust (RPM): } 1.357e - 06 \cdot \text{RPM}^2 + -1.630e - 03 \cdot \text{RPM} + 1.841e + 00$$

$$\text{Torque (RPM): } 3.076e - 08 \cdot \text{RPM}^2 + -4.079e - 05 \cdot \text{RPM} + 5.122e - 02$$

$$\text{Mechanical power (RPM): } \frac{2\pi \cdot \text{Torque}[\text{Nm}] \cdot \text{RPM}}{60}$$

$$\text{Formulas used to calculate FOM} : C_T = \frac{T}{\rho RPS^2 D^4} \quad C_P = \frac{P_{mech}}{\rho RPS^3 D^5} \quad FOM = \sqrt{\frac{2}{\pi} \frac{C_T^{3/2}}{C_P}}$$