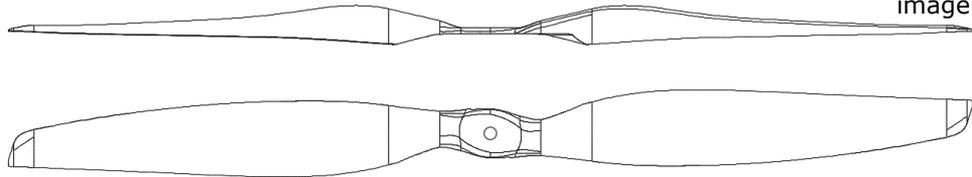


# PROPELLER

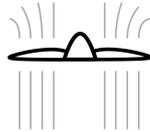
## 20x6.7 2B MC L

PN: 2200670, 2200671

\*Illustrative  
image only



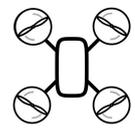
31 g  
Mass



15.0 kgf  
Max Thrust



20.0"  
Diameter



Multicopter

Engine/Motor type: Electric

Rotation direction: Counter-clockwise and Clockwise  
available (Direction Guide)

Mass [g]:  $31 \pm 10.0\%$

Moment of inertia [ $\text{kgm}^2$ ]<sup>1</sup>:  $6.67\text{e-}04$

Core diameter [mm]: 19.5 (Drilling guide)

Limit hover RPM<sup>2</sup>: 9000

Limit forward speed [m/s]: Not calculated

Working temperature [°C]: from -20°C to 60°C

Production method: Wet layup

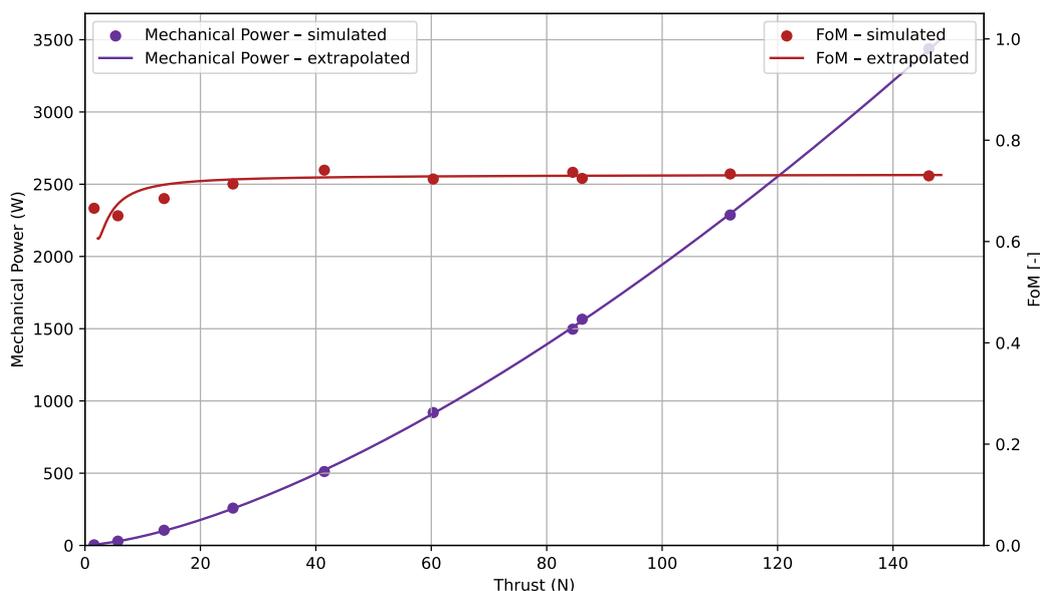
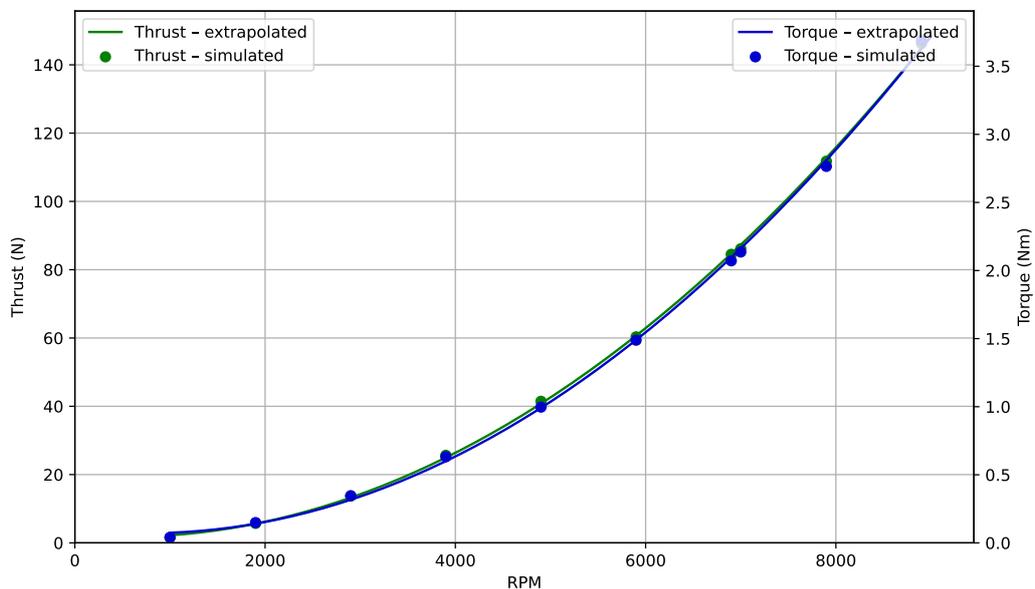
<sup>1</sup> Moment of inertia is only an estimation:  $I = \frac{1}{24} \cdot \text{mass} \cdot \text{diameter}^2 \cdot n^{\circ} \text{of blades}$

<sup>2</sup> RPM is limited by tip speed; forward speed reduces the limit.

For more information or custom propeller options, contact [info@mejlizk.eu](mailto:info@mejlizk.eu).  
Operation manual: Propeller Maintenance and Repair Manual

# Static Performance Data

Simulated data



$$\text{Thrust (RPM): } 2.043e - 06 \cdot \text{RPM}^2 + -2.170e - 03 \cdot \text{RPM} + 2.354e + 00$$

$$\text{Torque (RPM): } 5.379e - 08 \cdot \text{RPM}^2 + -8.221e - 05 \cdot \text{RPM} + 1.027e - 01$$

$$\text{Mechanical power (RPM): } \frac{2\pi \cdot \text{Torque}[\text{Nm}] \cdot \text{RPM}}{60}$$

Formulas used to calculate FOM :

$$C_T = \frac{T}{\rho RPS^2 D^4} \quad C_P = \frac{P_{mech}}{\rho RPS^3 D^5} \quad FOM = \sqrt{\frac{2}{\pi}} \frac{C_T^{3/2}}{C_P}$$