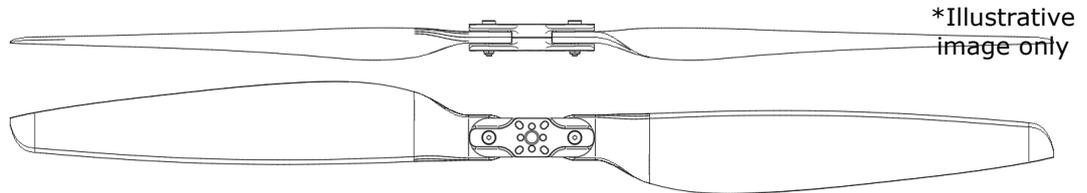


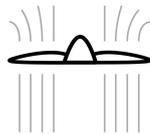
PROPELLER

22x7.4 2B MC gen.02 Folding

PN: 122000745, 122000746



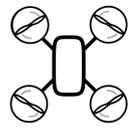
70 g
Mass



18.9 kgf
Max Thrust



22.0"
Diameter



Multicopter

Engine/Motor type: Electric

Rotation direction: Counter-clockwise and Clockwise available
(Direction Guide)

Features: Folding Blades

Mass [g]: $70 \pm 5.0\%$

Moment of inertia [kgm^2]¹: $5.12\text{e-}04$

Center hole diameter [mm]: 6

Drilling pattern [mm]: 4xM3 on 12 and 18-20

Limit hover RPM²: 8200

Limit forward speed [m/s]: 25

Working temperature [°C]: from -20°C to 110°C

Production method: Prepreg

Tests performed³: Overspin Test, Centrifugal Load Test, Tip Deflection Test

¹ Moment of inertia is measured.

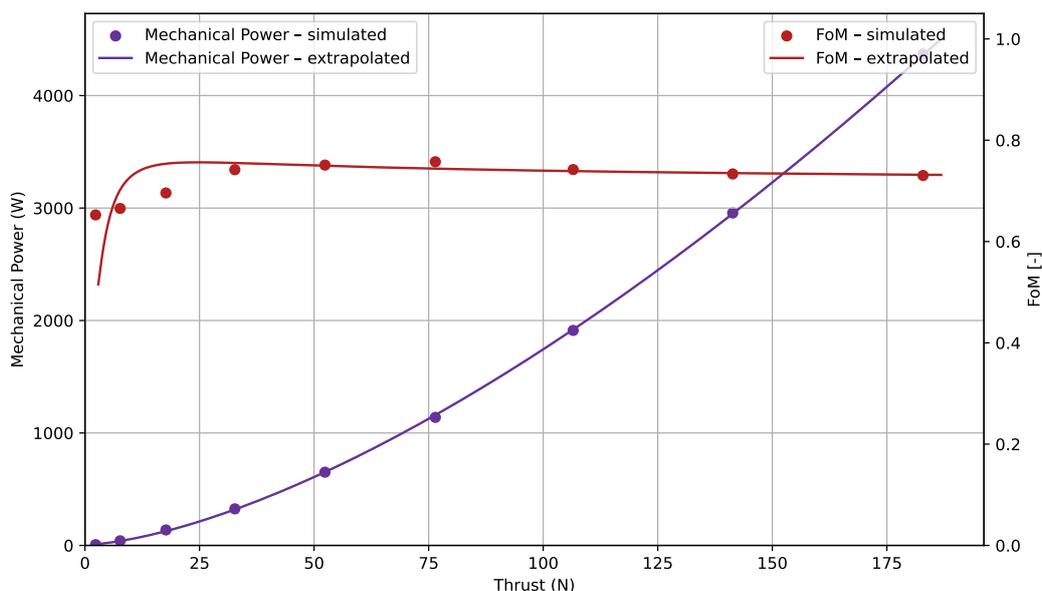
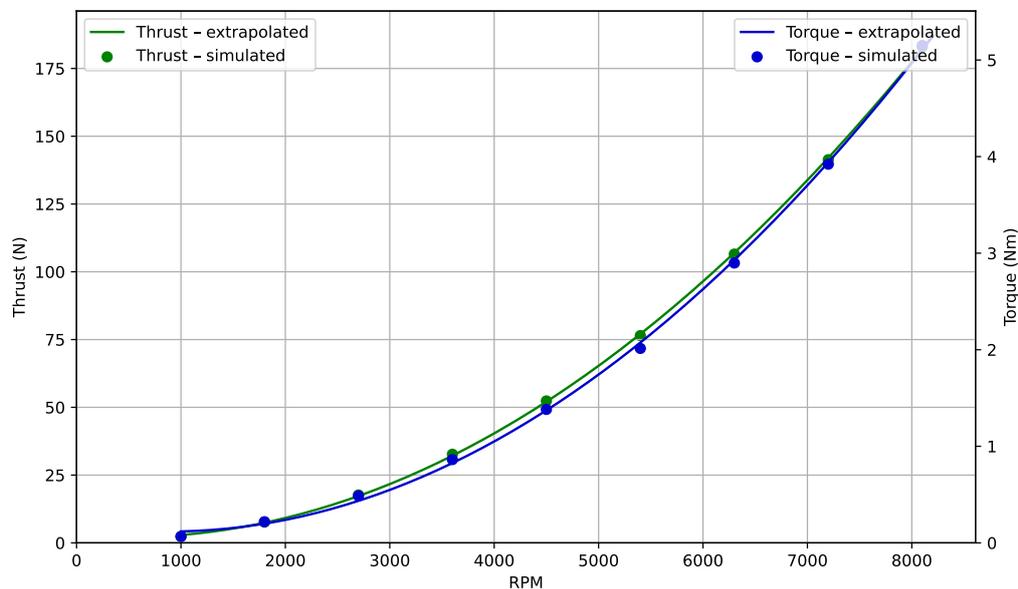
² RPM is limited by tip speed; forward speed reduces the limit.

³ Tests performed on a sample propeller.

For more information or custom propeller options, contact info@mejluk.eu.
Operation manual: Propeller Maintenance and Repair Manual

Static Performance Data

Simulated data



$$\text{Thrust (RPM): } 3.108e - 06 \cdot \text{RPM}^2 + -3.050e - 03 \cdot \text{RPM} + 2.846e + 00$$

$$\text{Torque (RPM): } 9.560e - 08 \cdot \text{RPM}^2 + -1.675e - 04 \cdot \text{RPM} + 1.905e - 01$$

$$\text{Mechanical power (RPM): } \frac{2\pi \cdot \text{Torque}[\text{Nm}] \cdot \text{RPM}}{60}$$

Formulas used to calculate FOM : $C_T = \frac{T}{\rho RPS^2 D^4}$ $C_P = \frac{P_{mech}}{\rho RPS^3 D^5}$ $FOM = \sqrt{\frac{2}{\pi} \frac{C_T^{3/2}}{C_P}}$