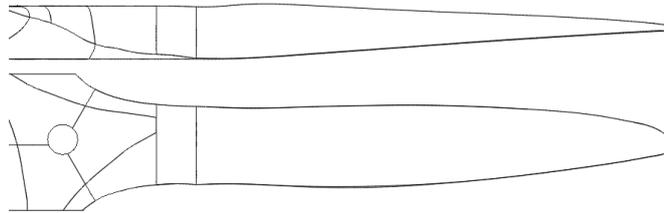


PROPELLER

28.5x12 3B GAS S

PN: 328623



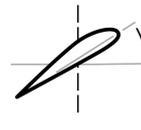
*Illustrative image only



328 g
Mass



28.5"
Diameter



12.0"
Pitch



Fixed wing

Engine/Motor type: Gas

Rotation direction: Counter-clockwise (Direction Guide)

Mass [g]: $328 \pm 10.0\%$

Moment of inertia [kgm²]¹: 2.15e-02

Core diameter [mm]: 54 (Drilling guide)

Limit RPM²: 6300

Working temperature [°C]: from -20°C to 60°C

Production method: Wet layup

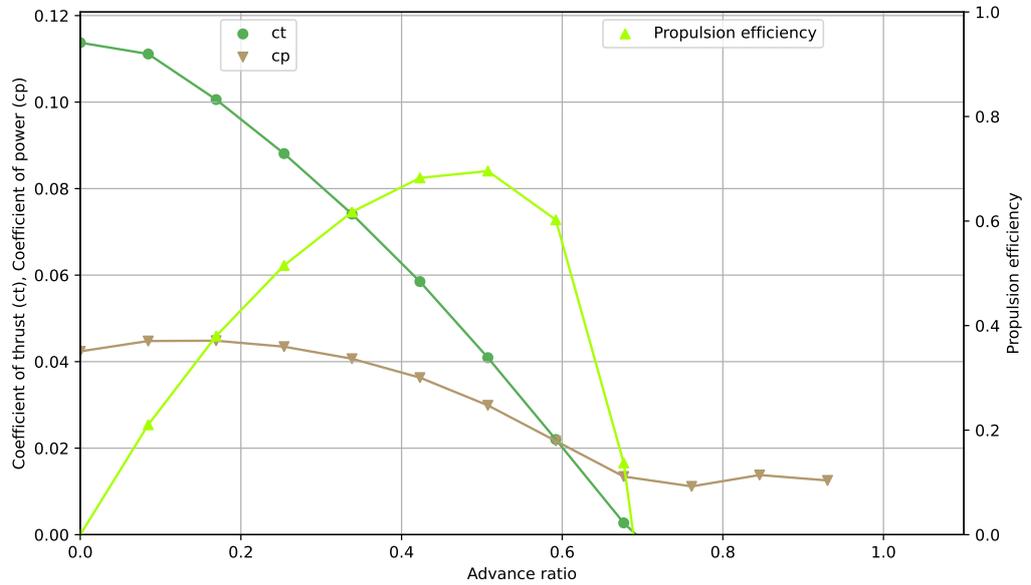
¹ Moment of inertia is only an estimation: $I = \frac{1}{24} \cdot mass \cdot diameter^2 \cdot n^{\circ} of blades$

² RPM is limited by tip speed; forward speed reduces the limit.

For more information or custom propeller options, contact info@mejzlik.eu.
Operation manual: Propeller Maintenance and Repair Manual

Dynamic Performance Data

Simulated data - at 4900 RPM



v_inf	Ct	Cp	Propulsion efficiency	Advance ratio
0.0	0.1137	0.0424	0.0	0.0
5.0	0.1111	0.0448	0.2103	0.0846
10.0	0.1006	0.0448	0.38	0.1692
15.0	0.0881	0.0435	0.5151	0.2537
20.0	0.0741	0.0407	0.6175	0.3383
25.0	0.0585	0.0363	0.6826	0.4229
30.0	0.0409	0.0299	0.6956	0.5075
35.0	0.022	0.0217	0.6023	0.592
40.0	0.0027	0.0134	0.1373	0.6766
45.0	-0.0121	0.0112	-0.829	0.7612
50.0	-0.019	0.0138	-1.1667	0.8458
55.0	-0.0282	0.0125	-2.096	0.9303

$$C_T = \frac{T}{\rho RPS^2 D^4}$$

$$C_P = \frac{P_{mech}}{\rho RPS^3 D^5}$$

$$\eta = \frac{C_T \cdot J}{C_P}$$

$$J = \frac{v_{inf}}{n \cdot D}$$