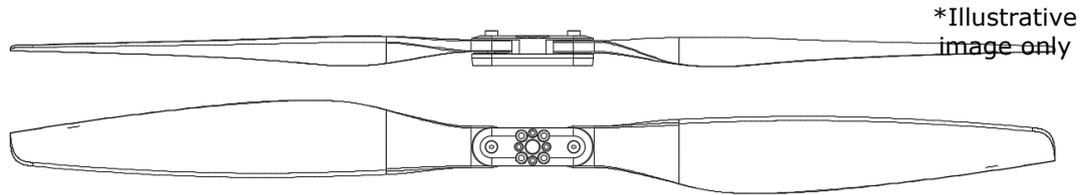


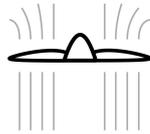
PROPELLER

30x10 2B MC Folding

PN: 23001004, 23001005, 23001006, 23001007



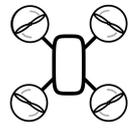
162 g
Mass



34.9 kgf
Max Thrust



30.0"
Diameter



Multicopter

Engine/Motor type: Electric

Rotation direction: Counter-clockwise and Clockwise available (Direction Guide)

Features: Folding Blades

Mass [g]: $162 \pm 10.0\%$

Moment of inertia [kgm^2]¹: $7.84\text{e-}03$

Center hole diameter [mm]: 10

Drilling pattern [mm]: -

Limit hover RPM²: 6000

Limit forward speed [m/s]: Not calculated

Working temperature [°C]: from -20°C to 60°C

Production method: Wet layup

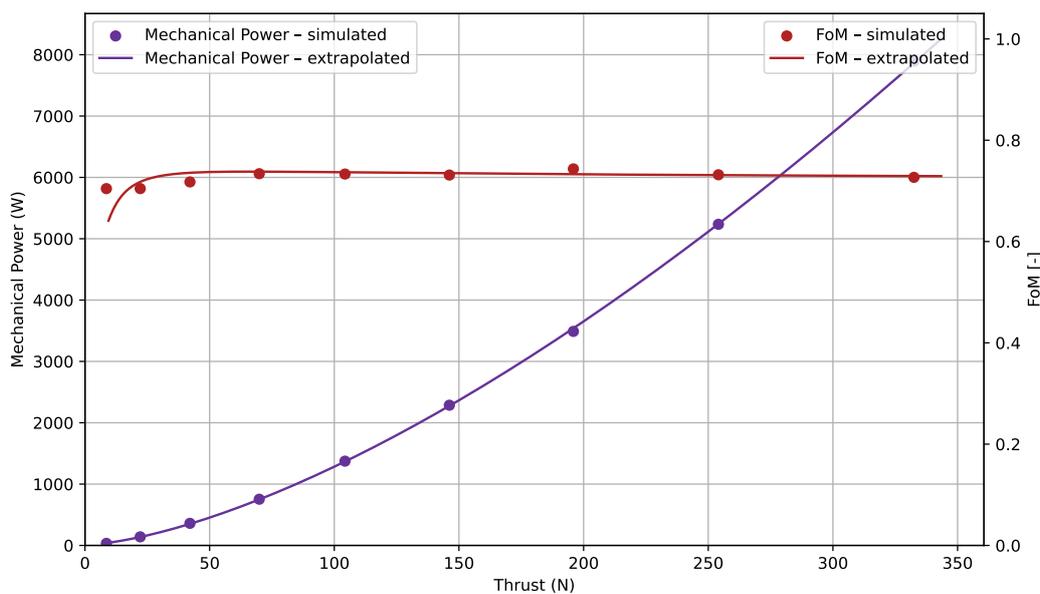
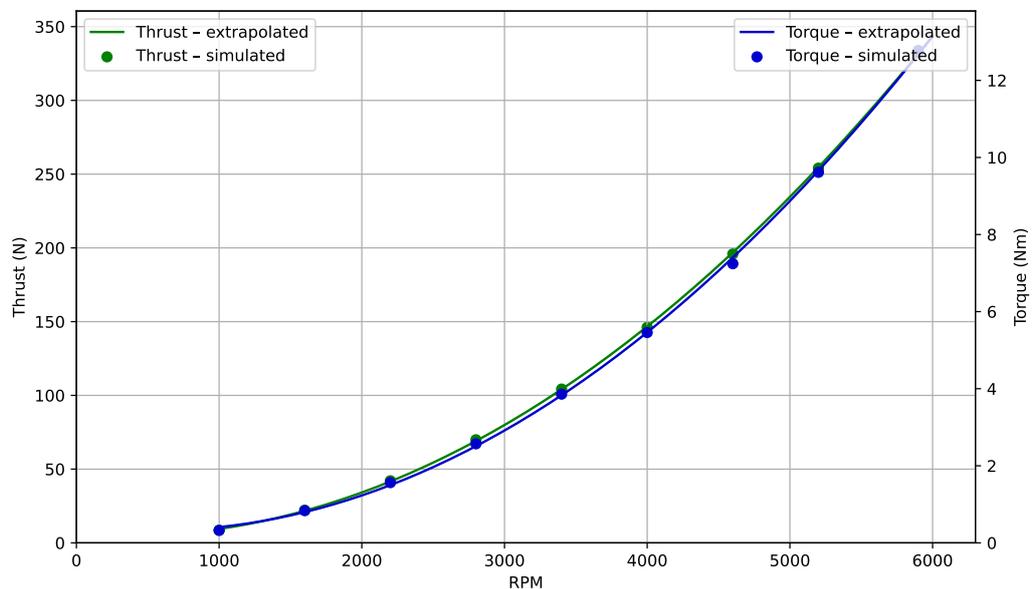
¹ Moment of inertia is only an estimation: $I = \frac{1}{24} \cdot mass \cdot diameter^2 \cdot n^{\circ}ofblades$

² RPM is limited by tip speed; forward speed reduces the limit.

For more information or custom propeller options, contact info@mejlzik.eu.
Operation manual: Propeller Maintenance and Repair Manual

Static Performance Data

Simulated data



$$\text{Thrust (RPM): } 1.052e - 05 \cdot \text{RPM}^2 + -6.850e - 03 \cdot \text{RPM} + 5.765e + 00$$

$$\text{Torque (RPM): } 4.311e - 07 \cdot \text{RPM}^2 + -4.712e - 04 \cdot \text{RPM} + 4.483e - 01$$

$$\text{Mechanical power (RPM): } \frac{2\pi \cdot \text{Torque}[\text{Nm}] \cdot \text{RPM}}{60}$$

Formulas used to calculate FOM :

$$C_T = \frac{T}{\rho RPS^2 D^4} \quad C_P = \frac{P_{mech}}{\rho RPS^3 D^5} \quad FOM = \sqrt{\frac{2}{\pi} \frac{C_T^{3/2}}{C_P}}$$