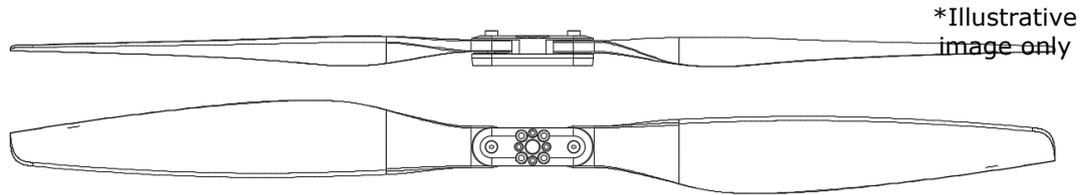


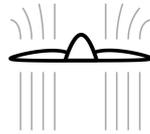
PROPELLER

32x10.6 2B MC Folding

PN: 23201060, 23201061, 23201062, 23201063



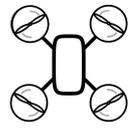
177 g
Mass



38.9 kgf
Max Thrust



32.0"
Diameter



Multicopter

Engine/Motor type: Electric

Rotation direction: Counter-clockwise and Clockwise available (Direction Guide)

Features: Folding Blades

Mass [g]: $177 \pm 10.0\%$

Moment of inertia [kgm^2]¹: $9.74\text{e-}03$

Center hole diameter [mm]: 10

Drilling pattern [mm]: -

Limit hover RPM²: 5600

Limit forward speed [m/s]: Not calculated

Working temperature [°C]: from -20°C to 60°C

Production method: Wet layup

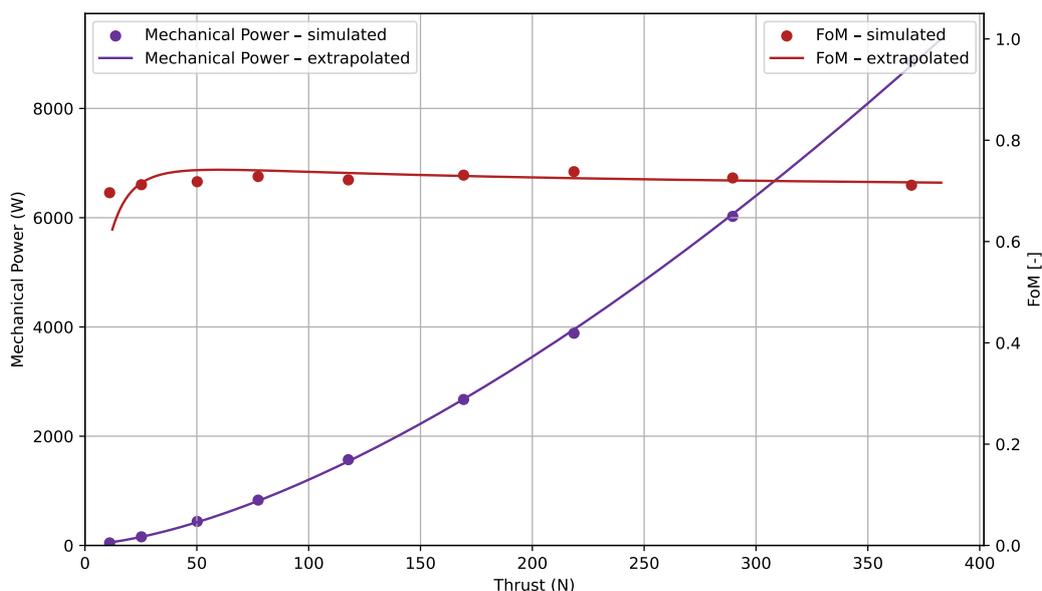
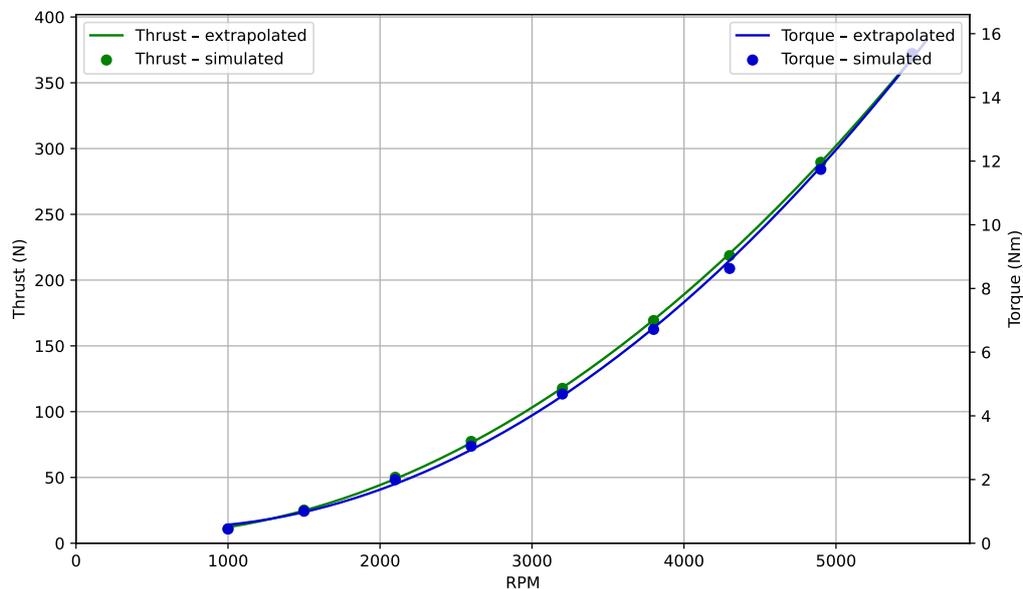
¹ Moment of inertia is only an estimation: $I = \frac{1}{24} \cdot mass \cdot diameter^2 \cdot n^0 \text{ of blades}$

² RPM is limited by tip speed; forward speed reduces the limit.

For more information or custom propeller options, contact info@mejlzik.eu.
Operation manual: Propeller Maintenance and Repair Manual

Static Performance Data

Simulated data



$$\text{Thrust (RPM): } 1.349e - 05 \cdot \text{RPM}^2 + -8.500e - 03 \cdot \text{RPM} + 7.236e + 00$$

$$\text{Torque (RPM): } 6.135e - 07 \cdot \text{RPM}^2 + -7.379e - 04 \cdot \text{RPM} + 7.048e - 01$$

$$\text{Mechanical power (RPM): } \frac{2\pi \cdot \text{Torque}[\text{Nm}] \cdot \text{RPM}}{60}$$

Formulas used to calculate FOM

$$C_T = \frac{T}{\rho RPS^2 D^4}$$

$$C_P = \frac{P_{mech}}{\rho RPS^3 D^5}$$

$$FOM = \sqrt{\frac{2}{\pi} \frac{C_T^{3/2}}{C_P}}$$