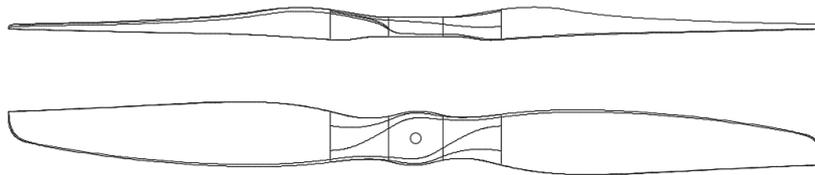


PROPELLER

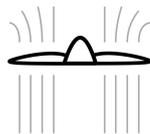
34x11.4 2B MC HD

PN: 23411400, 23411401

*Illustrative
image only



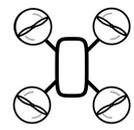
165 g
Mass



44.0 kgf
Max Thrust



34.0"
Diameter



Multicopter

Engine/Motor type: Electric

Rotation direction: Counter-clockwise and Clockwise
available (Direction Guide)

Mass [g]: $165 \pm 5.0\%$

Moment of inertia [kgm^2]¹: $1.03\text{e-}02$

Core diameter [mm]: 46 (Drilling guide)

Limit hover RPM²: 5300

Limit forward speed [m/s]: Not calculated

Working temperature [°C]: from -20°C to 60°C

Production method: Wet layup

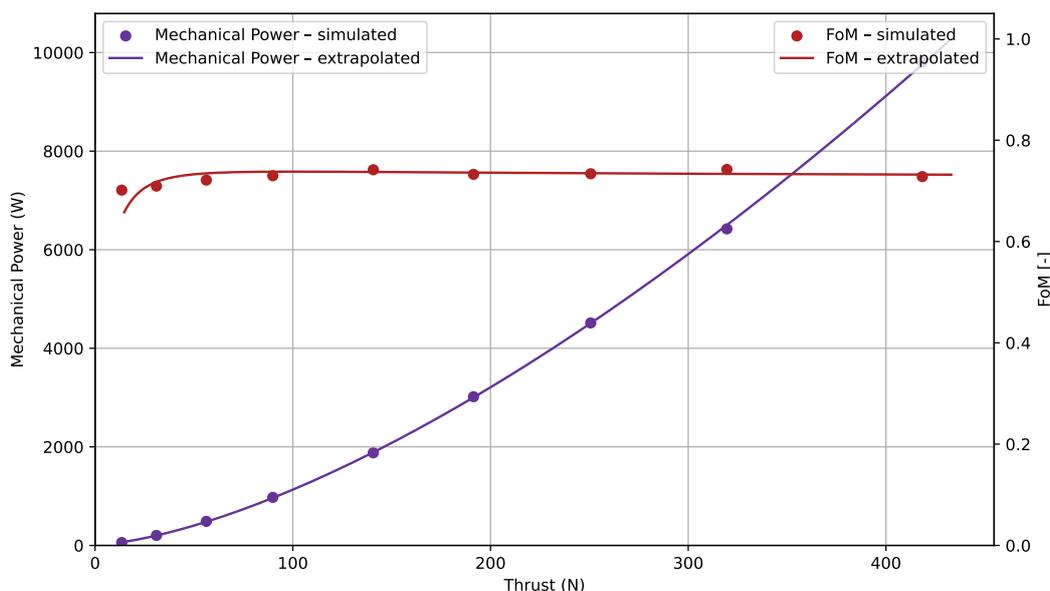
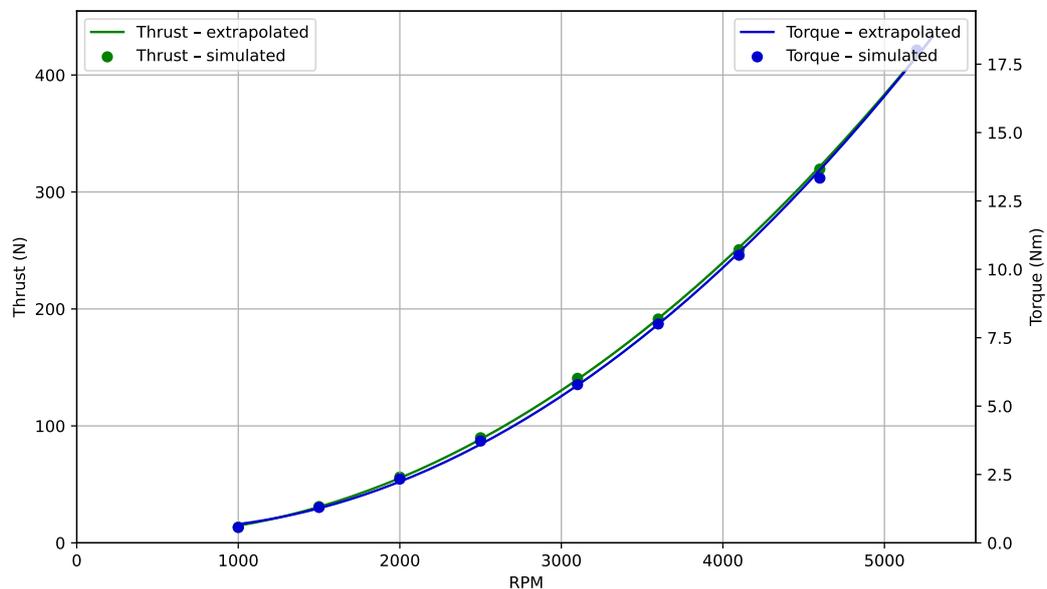
¹ Moment of inertia is only an estimation: $I = \frac{1}{24} \cdot \text{mass} \cdot \text{diameter}^2 \cdot n^{\circ} \text{ of blades}$

² RPM is limited by tip speed; forward speed reduces the limit.

For more information or custom propeller options, contact info@mejzlik.eu.
Operation manual: Propeller Maintenance and Repair Manual

Static Performance Data

Simulated data



$$\text{Thrust (RPM): } 1.715e - 05 \cdot \text{RPM}^2 + -1.079e - 02 \cdot \text{RPM} + 8.356e + 00$$

$$\text{Torque (RPM): } 7.861e - 07 \cdot \text{RPM}^2 + -8.052e - 04 \cdot \text{RPM} + 7.034e - 01$$

$$\text{Mechanical power (RPM): } \frac{2\pi \cdot \text{Torque}[\text{Nm}] \cdot \text{RPM}}{60}$$

Formulas used to calculate FOM : $C_T = \frac{T}{\rho RPS^2 D^4}$ $C_P = \frac{P_{mech}}{\rho RPS^3 D^5}$ $FOM = \sqrt{\frac{2}{\pi} \frac{C_T^{3/2}}{C_P}}$